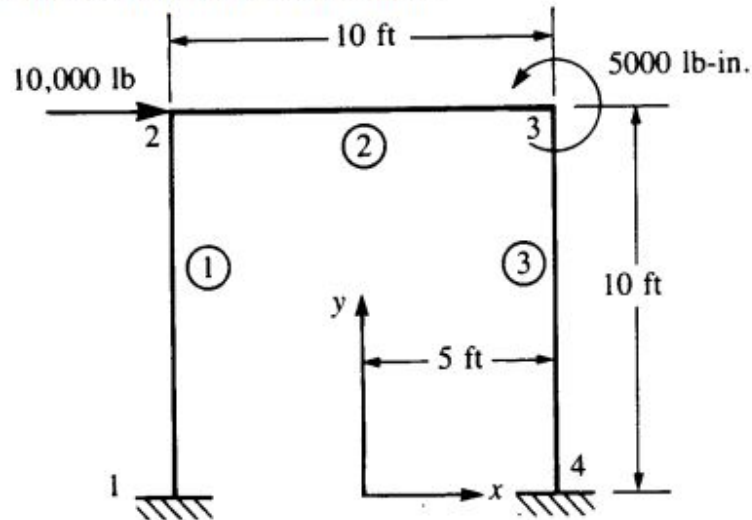


### Example #1

Consider the frame shown in the figure below.



The frame is fixed at nodes 1 and 4 and subjected to a positive horizontal force of 10,000 lb applied at node 2 and to a positive moment of 5,000 lb-in. applied at node 3. Let  $E = 30 \times 10^6$  psi and  $A = 10 \text{ in.}^2$  for all elements, and let  $I = 200 \text{ in.}^4$  for elements 1 and 3, and  $I = 100 \text{ in.}^4$  for element 2.

**Element 1:** The angle between  $x$  and  $\hat{x}$  is  $90^\circ$

$$C = 0 \quad S = 1$$

where

$$\frac{12I}{L^2} = \frac{12(200)}{(120)^2} = 0.167 \text{ in}^2 \quad \frac{6I}{L} = \frac{6(200)}{120} = 10.0 \text{ in}^3$$

$$\frac{E}{L} = \frac{30 \times 10^6}{120} = 250,000 \text{ lb/in}^3$$

Therefore, for element 1:

$$k^{(1)} = 250,000 \begin{bmatrix} d_{1x} & d_{1y} & \phi_1 & d_{2x} & d_{2y} & \phi_2 \\ 0.167 & 0 & -10 & -0.167 & 0 & -10 \\ 0 & 10 & 0 & 0 & -10 & 0 \\ -10 & 0 & 800 & 10 & 0 & 400 \\ -0.167 & 0 & 10 & 0.167 & 0 & 10 \\ 0 & -10 & 0 & 0 & 10 & 0 \\ -10 & 0 & 400 & 10 & 0 & 800 \end{bmatrix} \text{ lb/in}$$

**Element 2:** The angle between  $x$  and  $\hat{x}$  is  $0^\circ$

$$C = 1 \quad S = 0$$

$$\frac{12I}{L^2} = \frac{12(100)}{(120)^2} = 0.0835 \text{ in}^2 \quad \frac{6I}{L} = \frac{6(100)}{120} = 5.0 \text{ in}^3$$

Therefore, for element 2:

$$k^{(2)} = 250,000 \begin{bmatrix} d_{2x} & d_{2y} & \phi_2 & d_{3x} & d_{3y} & \phi_3 \\ 10 & 0 & 0 & -10 & 0 & 0 \\ 0 & 0.0835 & 5 & 0 & 0.0835 & 5 \\ 0 & 5 & 400 & 0 & -5 & 200 \\ -10 & 0 & 0 & 10 & 0 & 0 \\ 0 & 0.0835 & -5 & 0 & 0.0835 & -5 \\ 0 & 5 & 200 & 0 & -5 & 400 \end{bmatrix} \text{ lb/in}$$

**Element 3:** The angle between  $x$  and  $\hat{x}$  is  $270^\circ$

$$C = 0 \quad S = -1$$

$$\frac{12I}{L^2} = \frac{12(200)}{(120)^2} = 0.167 \text{ in}^2 \quad \frac{6I}{L} = \frac{6(200)}{120} = 10.0 \text{ in}^3$$

$$\frac{E}{L} = \frac{30 \times 10^6}{120} = 250,000 \text{ lb/in}^3$$

Therefore, for element 3:

$$k^{(3)} = 250,000 \begin{bmatrix} d_{3x} & d_{3y} & \phi_3 & d_{4x} & d_{4y} & \phi_4 \\ 0.167 & 0 & 10 & -0.167 & 0 & 10 \\ 0 & 10 & 0 & 0 & -10 & 0 \\ 10 & 0 & 800 & -10 & 0 & 400 \\ -0.167 & 0 & -10 & 0.167 & 0 & -10 \\ 0 & -10 & 0 & 0 & 10 & 0 \\ 10 & 0 & 400 & -10 & 0 & 800 \end{bmatrix} \text{ lb/in}$$

The boundary conditions for this problem are:

$$d_{1x} = d_{1y} = \phi_1 = d_{2x} = d_{2y} = \phi_2 = 0$$

After applying the boundary conditions the global beam equations reduce to:

$$\begin{Bmatrix} 10,000 \\ 0 \\ 0 \\ 0 \\ 0 \\ 5,000 \end{Bmatrix} = 2.5 \times 10^5 \begin{bmatrix} 10.167 & 0 & 10 & -10 & 0 & 0 \\ 0 & 10.0835 & 5 & 0 & -0.0835 & 5 \\ 10 & 5 & 1200 & 0 & -5 & 200 \\ -10 & 0 & 0 & 10.167 & 0 & 10 \\ 0 & -0.0835 & -5 & 0 & 10.0835 & -5 \\ 0 & 5 & 200 & 10 & -5 & 1200 \end{bmatrix} \begin{Bmatrix} d_{2x} \\ d_{2y} \\ \phi_2 \\ d_{3x} \\ d_{3y} \\ \phi_3 \end{Bmatrix}$$

Solving the above equations gives:

$$\begin{Bmatrix} d_{2x} \\ d_{2y} \\ \phi_2 \\ d_{3x} \\ d_{3y} \\ \phi_3 \end{Bmatrix} = \begin{Bmatrix} 0.211 \text{ in} \\ 0.00148 \text{ in} \\ -0.00153 \text{ rad} \\ 0.209 \text{ in} \\ -0.00148 \text{ in} \\ -0.00149 \text{ rad} \end{Bmatrix}$$

**Element 1:** The element force-displacement equations can be obtained using  $\hat{f} = \hat{k}\bar{T}d$ . Therefore,  $\bar{T}d$  is:

$$\bar{T}d = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} d_{1x} = 0 \\ d_{1y} = 0 \\ \phi_1 = 0 \\ d_{2x} = 0.211 \text{ in} \\ d_{2y} = 0.00148 \text{ in} \\ \phi_2 = -0.00153 \text{ rad} \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ 0.00148 \text{ in} \\ -0.211 \text{ in} \\ -0.00153 \text{ rad} \end{Bmatrix}$$

Recall the elemental stiffness matrix is:

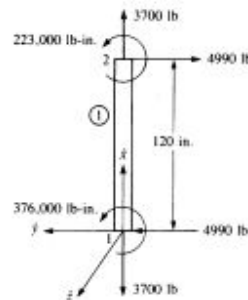
$$\hat{k} = \begin{bmatrix} C_1 & 0 & 0 & -C_1 & 0 & 0 \\ 0 & 12C_2 & 6LC_2 & 0 & -12C_2 & 6LC_2 \\ 0 & 6LC_2 & 4C_2L^2 & 0 & -6LC_2 & 2C_2L^2 \\ -C_1 & 0 & 0 & C_1 & 0 & 0 \\ 0 & -12C_2 & -6LC_2 & 0 & 12C_2 & -6LC_2 \\ 0 & 6LC_2 & 2C_2L^2 & 0 & -6LC_2 & 4C_2L^2 \end{bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}^{(1)} = \hat{k}\bar{T}d = 2.5 \times 10^5 \begin{bmatrix} 10 & 0 & 0 & -10 & 0 & 0 \\ 0 & 0.167 & 10 & 0 & -0.167 & 10 \\ 0 & 10 & 800 & 0 & -10 & 400 \\ -10 & 0 & 0 & 10 & 0 & 10 \\ 0 & -0.167 & -10 & 0 & 0.167 & -10 \\ 0 & 10 & 400 & 0 & -10 & 800 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ 0 \\ 0.00148 \text{ in} \\ -0.211 \text{ in} \\ -0.00153 \text{ rad} \end{Bmatrix}$$

Simplifying the above equations gives:

$$\begin{Bmatrix} \hat{f}_{1x} \\ \hat{f}_{1y} \\ \hat{m}_1 \\ \hat{f}_{2x} \\ \hat{f}_{2y} \\ \hat{m}_2 \end{Bmatrix} = \begin{Bmatrix} -3,700 \text{ lb} \\ 4,990 \text{ lb} \\ 376 \text{ k} \cdot \text{in} \\ 3,700 \text{ lb} \\ -4,990 \text{ lb} \\ 223 \text{ k} \cdot \text{in} \end{Bmatrix}$$



**Element 2:** The element force-displacement equations are:

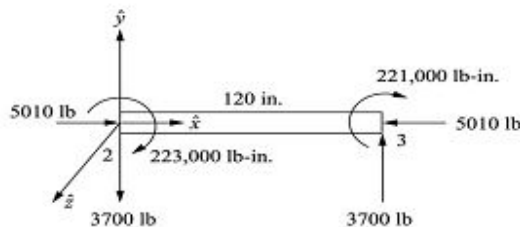
$$\bar{T}d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} d_{2x} = 0.211 \text{ in} \\ d_{2y} = 0.00148 \text{ in} \\ \phi_2 = -0.00153 \text{ rad} \\ d_{3x} = 0.209 \text{ in} \\ d_{3y} = -0.00148 \text{ in} \\ \phi_3 = -0.00149 \text{ rad} \end{Bmatrix} = \begin{Bmatrix} -0.211 \text{ in} \\ 0.00148 \text{ in} \\ -0.00153 \text{ rad} \\ 0.209 \text{ in} \\ -0.00148 \text{ in} \\ -0.00149 \text{ rad} \end{Bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}^{(2)} = \hat{k}\bar{T}d = 2.5 \times 10^5 \begin{bmatrix} 10 & 0 & 0 & -10 & 0 & 0 \\ 0 & 0.0833 & 5 & 0 & -0.0833 & 5 \\ 0 & 5 & 400 & 0 & -5 & 200 \\ -10 & 0 & 0 & 10 & 0 & 0 \\ 0 & -0.0833 & -5 & 0 & 0.0833 & -5 \\ 0 & 5 & 200 & 0 & -5 & 400 \end{bmatrix} \begin{Bmatrix} 0.211 \text{ in} \\ 0.00148 \text{ in} \\ -0.00153 \text{ rad} \\ 0.209 \text{ in} \\ -0.00148 \text{ in} \\ -0.00149 \text{ rad} \end{Bmatrix}$$

Simplifying the above equations gives:

$$\begin{Bmatrix} \hat{f}_{2x} \\ \hat{f}_{2y} \\ \hat{m}_2 \\ \hat{f}_{3x} \\ \hat{f}_{3y} \\ \hat{m}_3 \end{Bmatrix} = \begin{Bmatrix} 5,010 \text{ lb} \\ -3,700 \text{ lb} \\ -223 \text{ k} \cdot \text{in} \\ -5,010 \text{ lb} \\ 3,700 \text{ lb} \\ -221 \text{ k} \cdot \text{in} \end{Bmatrix}$$



**Element 3:** The element force-displacement equations are:

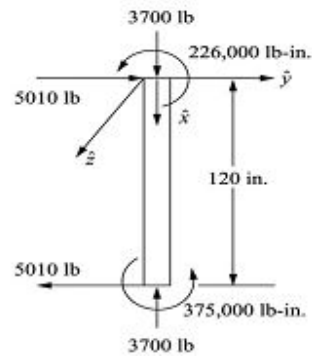
$$\bar{T}d = \begin{bmatrix} 0 & -1 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} d_{3x} = 0.209 \text{ in} \\ d_{3y} = -0.00148 \text{ in} \\ \phi_3 = -0.00149 \text{ rad} \\ d_{4x} = 0 \\ d_{4y} = 0 \\ \phi_4 = 0 \end{bmatrix} = \begin{bmatrix} 0.00148 \text{ in} \\ 0.209 \text{ in} \\ -0.00149 \text{ rad} \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}^{(3)} = \hat{k}\bar{T}d = 2.5 \times 10^5 \begin{bmatrix} 10 & 0 & 0 & -10 & 0 & 0 \\ 0 & 0.167 & 10 & 0 & -0.167 & 10 \\ 0 & 10 & 800 & 0 & -10 & 400 \\ -10 & 0 & 0 & 10 & 0 & 10 \\ 0 & -0.167 & -10 & 0 & 0.167 & -10 \\ 0 & 10 & 400 & 0 & -10 & 800 \end{bmatrix} \begin{bmatrix} 0.00148 \text{ in} \\ 0.209 \text{ in} \\ -0.00149 \text{ rad} \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

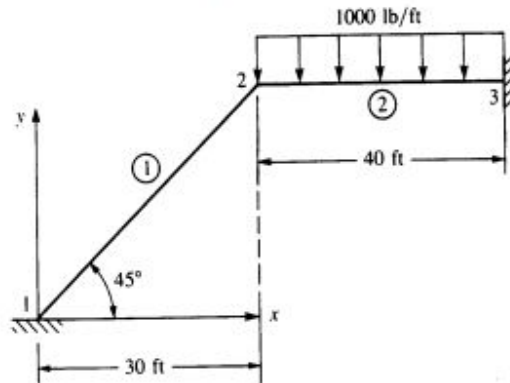
Simplifying the above equations gives:

$$\begin{Bmatrix} \hat{f}_{3x} \\ \hat{f}_{3y} \\ \hat{m}_3 \\ \hat{f}_{4x} \\ \hat{f}_{4y} \\ \hat{m}_4 \end{Bmatrix} = \begin{Bmatrix} 3,700 \text{ lb} \\ 5,010 \text{ lb} \\ 226 \text{ k} \cdot \text{in} \\ -3,700 \text{ lb} \\ -5,010 \text{ lb} \\ 375 \text{ k} \cdot \text{in} \end{Bmatrix}$$



## Example # 2

Consider the frame shown in the figure below.



The frame is fixed at nodes 1 and 3 and subjected to a positive distributed load of 1,000 lb/ft applied along element 2. Let  $E = 30 \times 10^6$  psi and  $A = 100 \text{ in.}^2$  for all elements, and let  $I = 1,000 \text{ in.}^4$  for all elements.

First we need to replace the distributed load with a set of equivalent nodal forces and moments acting at nodes 2 and 3. For a beam with both end fixed, subjected to a uniform distributed load,  $w$ , the nodal forces and moments are:

$$f_{2y} = f_{3y} = -\frac{wL}{2} = -\frac{(1,000 \text{ lb/ft})40 \text{ ft}}{2} = -20 \text{ k}$$

$$m_2 = -m_3 = -\frac{wL^2}{12} = -\frac{(1,000 \text{ lb/ft})(40 \text{ ft})^2}{12} = -133,333 \text{ lb} \cdot \text{ft} = 1,600 \text{ k} \cdot \text{in}$$

If we consider only the parts of the stiffness matrix associated with the three degrees of freedom at node 2, we get:

**Element 1:** The angle between  $x$  and  $\hat{x}$  is  $45^\circ$

$$C = 0.707 \quad S = 0.707$$

where

$$\frac{E}{L} = \frac{30 \times 10^6}{509} = 58.93 \text{ k/in}^3 \quad \frac{12I}{L^2} = \frac{12(1,000)}{(12 \times 30\sqrt{2})^2} = 0.0463 \text{ in}^2$$

$$\frac{6I}{L} = \frac{6(1,000)}{12 \times 30\sqrt{2}} = 11.78551 \text{ in}^3$$

Therefore, for element 1:

$$k^{(1)} = 58.93 \begin{bmatrix} d_{2x} & d_{2y} & \phi_2 \\ 50.02 & 49.98 & 8.33 \\ 49.98 & 50.02 & -8.33 \\ 8.33 & -8.33 & 4000 \end{bmatrix} \text{ k/in}$$

Simplifying the above equation:

$$k^{(1)} = \begin{matrix} & d_{2x} & d_{2y} & \phi_2 \\ \begin{bmatrix} 2,948 & 2,945 & 491 \\ 2,945 & 2,948 & -491 \\ 491 & -491 & 235,700 \end{bmatrix} & & & \end{matrix} \left. \vphantom{\begin{matrix} d_{2x} \\ d_{2y} \\ \phi_2 \end{matrix}} \right\} k/in$$

**Element 2:** The angle between  $x$  and  $x$  is  $0^\circ$

$$C = 1 \quad S = 0$$

where

$$\frac{E}{L} = \frac{30 \times 10^8}{480} = 62.5 \text{ k/in}^3 \quad \frac{12I}{L^2} = \frac{12(1,000)}{(12 \times 40)^2} = 0.0521 \text{ in}^2$$

$$\frac{6I}{L} = \frac{6(1,000)}{12 \times 40} = 12.5 \text{ in}^3$$

Therefore, for element 2:

$$k^{(2)} = 62.50 \begin{matrix} & d_{2x} & d_{2y} & \phi_2 \\ \begin{bmatrix} 100 & 0 & 0 \\ 0 & 0.052 & 12.5 \\ 0 & 12.5 & 4,000 \end{bmatrix} & & & \end{matrix} \left. \vphantom{\begin{matrix} d_{2x} \\ d_{2y} \\ \phi_2 \end{matrix}} \right\} k/in$$

Simplifying the above equation:

$$k^{(2)} = \begin{matrix} & d_{2x} & d_{2y} & \phi_2 \\ \begin{bmatrix} 6,250 & 0 & 0 \\ 0 & 3.25 & 781.25 \\ 0 & 781.25 & 250,000 \end{bmatrix} & & & \end{matrix} \left. \vphantom{\begin{matrix} d_{2x} \\ d_{2y} \\ \phi_2 \end{matrix}} \right\} k/in$$

The global beam equations reduce to:

$$\begin{Bmatrix} 0 \\ -20 \text{ k} \\ -1,600 \text{ k} \cdot \text{in} \end{Bmatrix} = \begin{bmatrix} 9,198 & 2,945 & 491 \\ 2,945 & 2,951 & 290 \\ 491 & 290 & 485,700 \end{bmatrix} \begin{Bmatrix} d_{2x} \\ d_{2y} \\ \phi_2 \end{Bmatrix}$$

Solving the above equations gives:

$$\begin{Bmatrix} d_{2x} \\ d_{2y} \\ \phi_2 \end{Bmatrix} = \begin{Bmatrix} 0.0033 \text{ in} \\ -0.0097 \text{ in} \\ -0.0033 \text{ rad} \end{Bmatrix}$$

**Element 1:** The element force-displacement equations can be obtained using  $\hat{f} = \hat{k}\bar{T}d$ . Therefore,  $\bar{T}d$  is:

$$\bar{T}d = \begin{bmatrix} 0.707 & 0.707 & 0 & 0 & 0 & 0 \\ -0.707 & 0.707 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0.707 & 0.707 & 0 \\ 0 & 0 & 0 & -0.707 & 0.707 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ 0 \\ 0.0033 \text{ in} \\ -0.0097 \text{ in} \\ -0.0033 \text{ rad} \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ -0.00452 \text{ in} \\ -0.0092 \text{ in} \\ -0.0033 \text{ rad} \end{Bmatrix}$$

Recall the elemental stiffness matrix is a function of values  $C_1$ ,  $C_2$ , and  $L$

$$C_1 = \frac{AE}{L} = \frac{(100)30 \times 10^5}{12 \times 30\sqrt{2}} = 5,893 \text{ } \%$$

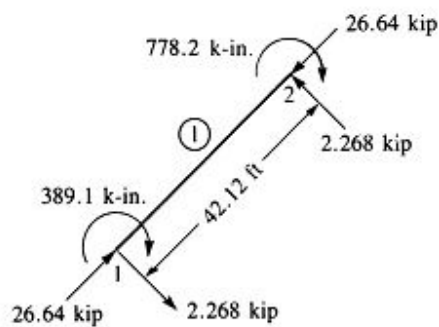
$$C_2 = \frac{EI}{L^3} = \frac{30 \times 10^5 (1,000)}{(12 \times 30\sqrt{2})^3} = 0.2273 \text{ } \%$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(1)} = \hat{k}\bar{T}d = \begin{bmatrix} 5,893 & 0 & 10 & -5,893 & 0 & 0 \\ 0 & 2,730 & 694.8 & 0 & -2,730 & 694.8 \\ 10 & 694.8 & 117,900 & 0 & -694.8 & 117,000 \\ -5,893 & 0 & 0 & 5,893 & 0 & 0 \\ 0 & -2,730 & -694.8 & 0 & 2,730 & -694.8 \\ 0 & 694.8 & 117,000 & 0 & -694.8 & 235,800 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ 0 \\ -0.00452 \text{ in} \\ -0.0092 \text{ in} \\ -0.0033 \text{ rad} \end{Bmatrix}$$

Simplifying the above equations gives:

$$\begin{Bmatrix} \hat{f}_{1x} \\ \hat{f}_{1y} \\ \hat{m}_1 \\ \hat{f}_{2x} \\ \hat{f}_{2y} \\ \hat{m}_2 \end{Bmatrix} = \begin{Bmatrix} 26.64 \text{ k} \\ -2.268 \text{ k} \\ -389.1 \text{ k} \cdot \text{in} \\ -26.64 \text{ k} \\ 2.268 \text{ k} \\ -778.2 \text{ k} \cdot \text{in} \end{Bmatrix}$$





**Element 2:** The element force-displacement equations are:

$$\bar{T}d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} 0.0033 \text{ in} \\ -0.0097 \text{ in} \\ -0.0033 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0.0033 \text{ in} \\ -0.0097 \text{ in} \\ -0.0033 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Recall the elemental stiffness matrix is a function of values  $C_1$ ,  $C_2$ , and  $L$

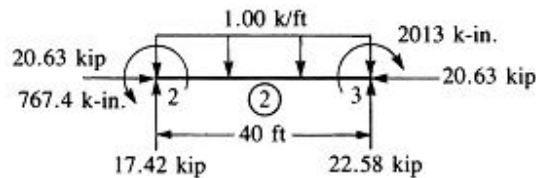
$$C_1 = \frac{AE}{L} = \frac{(100)30 \times 10^6}{12 \times 40} = 6,250 \text{ } \frac{\text{lb}}{\text{in}} \quad C_2 = \frac{EI}{L^3} = \frac{30 \times 10^6(1,000)}{(12 \times 40)^3} = 0.2713 \text{ } \frac{\text{lb}}{\text{in}^2}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(2)} = \hat{k}\bar{T}d = \begin{bmatrix} 6,250 & 0 & 0 & -6,250 & 0 & 0 \\ 0 & 3.25 & 781.1 & 0 & -3.25 & 781.1 \\ 0 & 781.1 & 250,000 & 0 & -781.1 & 125,000 \\ -6,250 & 0 & 0 & 6,250 & 0 & 0 \\ 0 & -3.25 & -781.1 & 0 & 3.25 & -781.1 \\ 0 & 781.1 & 125,000 & 0 & -781.1 & 250,000 \end{bmatrix} \begin{Bmatrix} -0.0033 \text{ in} \\ -0.0097 \text{ in} \\ -0.0033 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Simplifying the above equations gives:

$$\hat{k}\hat{d} = \begin{Bmatrix} 20.63 \text{ k} \\ -2.58 \text{ k} \\ -832.57 \text{ k} \cdot \text{in} \\ -20.63 \text{ k} \\ 2.58 \text{ k} \\ -412.50 \text{ k} \cdot \text{in} \end{Bmatrix}$$

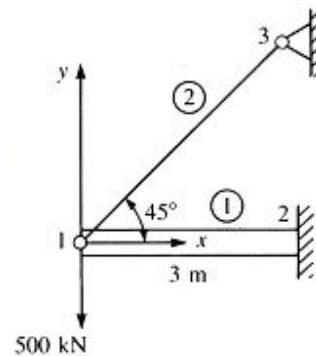


To obtain the actual element local forces, we must subtract the equivalent nodal forces.

$$\begin{Bmatrix} \hat{f}_{2x} \\ \hat{f}_{2y} \\ \hat{m}_2 \\ \hat{f}_{3x} \\ \hat{f}_{3y} \\ \hat{m}_3 \end{Bmatrix} = \begin{Bmatrix} 20.63 \text{ k} \\ -2.58 \text{ k} \\ -832.57 \text{ k} \cdot \text{in} \\ -20.63 \text{ k} \\ 2.58 \text{ k} \\ -412.50 \text{ k} \cdot \text{in} \end{Bmatrix} - \begin{Bmatrix} 0 \\ -20 \text{ k} \\ -1600 \text{ k} \cdot \text{in} \\ 0 \\ -20 \text{ k} \\ 1600 \text{ k} \cdot \text{in} \end{Bmatrix} = \begin{Bmatrix} 20.63 \text{ k} \\ 17.42 \text{ k} \\ 767.4 \text{ k} \cdot \text{in} \\ -20.63 \text{ k} \\ 22.58 \text{ k} \\ -2,013 \text{ k} \cdot \text{in} \end{Bmatrix}$$

### Example # 3

The frame shown on the right is fixed at nodes 2 and 3 and subjected to a concentrated load of 500 kN applied at node 1. For the bar,  $A = 1 \times 10^{-3} \text{ m}^2$ , for the beam,  $A = 2 \times 10^{-3} \text{ m}^2$ ,  $I = 5 \times 10^{-5} \text{ m}^4$ , and  $L = 3 \text{ m}$ . Let  $E = 210 \text{ GPa}$  for both elements.



**Beam Element 1:** The angle between  $x$  and  $\hat{x}$  is  $0^\circ$

$$C = 1 \quad S = 0$$

where

$$\frac{12I}{L^2} = \frac{12(5 \times 10^{-5})}{(3)^2} = 6.67 \times 10^{-5} \text{ m}^2 \quad \frac{6I}{L} = \frac{6(5 \times 10^{-5})}{3} = 10^{-4} \text{ m}^3$$

$$\frac{E}{L} = \frac{210 \times 10^9}{3} = 70 \times 10^9 \text{ kN/m}^3$$

Therefore, for element 1:

$$k^{(1)} = 70 \times 10^9 \begin{bmatrix} a_{1x} & a_{1y} & \phi_1 \\ 2 & 0 & 0 \\ 0 & 0.067 & 0.10 \\ 0 & 0.10 & 0.20 \end{bmatrix} \text{ kN/m}$$

**Bar Element 2:** The angle between  $x$  and  $\hat{x}$  is  $45^\circ$

$$C = 0.707 \quad S = 0.707$$

where

$$k^{(2)} = \frac{10^{-3} \text{ m}^2 (210 \times 10^9 \text{ kN/m}^2)}{4.24 \text{ m}} \begin{bmatrix} a_{2x} & a_{2y} \\ 0.5 & 0.5 \\ 0.5 & 0.5 \end{bmatrix} \text{ kN/m}$$

$$k^{(2)} = 70 \times 10^9 \begin{bmatrix} a_{2x} & a_{2y} \\ 0.354 & 0.354 \\ 0.354 & 0.354 \end{bmatrix} \text{ kN/m}$$

Assembling the elemental stiffness matrices we obtain the global stiffness matrix

$$K = 70 \times 10^9 \begin{bmatrix} 2.354 & 0.354 & 0 \\ 0.354 & 0.421 & 0.10 \\ 0 & 0.10 & 0.20 \end{bmatrix} \text{ kN/m}$$

The global equations are:

$$\begin{Bmatrix} 0 \\ -500 \text{ kN} \\ 0 \end{Bmatrix} = 70 \times 10^3 \text{ kN/m} \begin{bmatrix} 2.354 & 0.354 & 0 \\ 0.354 & 0.421 & 0.10 \\ 0 & 0.10 & 0.20 \end{bmatrix} \begin{Bmatrix} d_x \\ d_y \\ \phi \end{Bmatrix}$$

Solving the above equations gives:

$$\begin{Bmatrix} d_x \\ d_y \\ \phi \end{Bmatrix} = \begin{Bmatrix} 0.00388 \text{ m} \\ -0.0225 \text{ m} \\ 0.0113 \text{ rad} \end{Bmatrix}$$

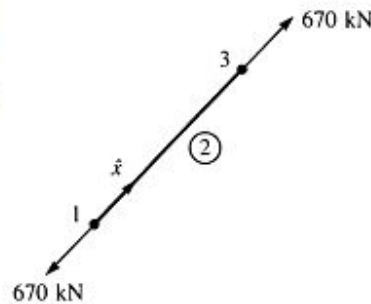
**Bar Element:** The bar element force-displacement equations can be obtained using  $\hat{f} = \hat{k}\bar{d}$ .

$$\begin{Bmatrix} \hat{f}_x \\ \hat{f}_{3x} \end{Bmatrix} = \frac{AE}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} C & S & 0 & 0 \\ 0 & 0 & C & S \end{bmatrix} \begin{Bmatrix} d_x \\ d_y \\ d_{3x} \\ d_{3y} \end{Bmatrix}$$

Therefore, the forces in the bar element are:

$$\hat{f}_x = \frac{AE}{L}(Cd_x + Sd_y) = -670 \text{ kN}$$

$$\hat{f}_{3x} = -\frac{AE}{L}(Cd_x + Sd_y) = 670 \text{ kN}$$



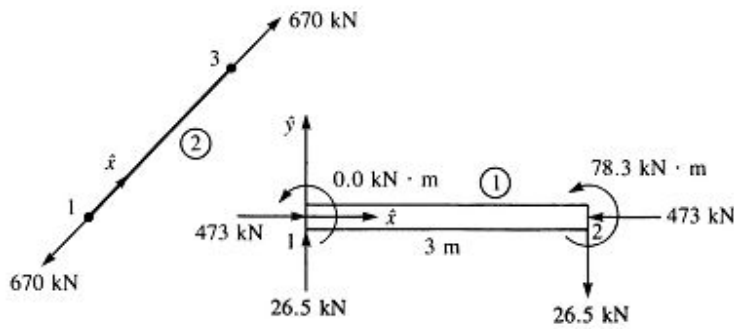
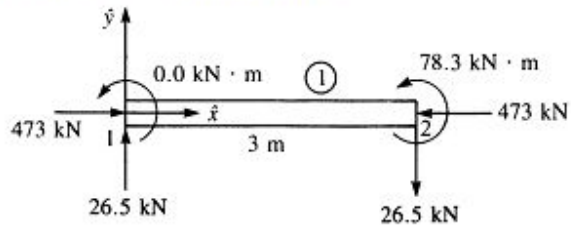
**Beam Element:** The beam element force-displacement equations can be obtained using  $\hat{f} = \hat{k}\bar{d}$ . Since the local axis coincides with the global coordinate system, and the displacements at node 2 are zero. Therefore, the local force-displacement equations are:

$$\hat{k} = \begin{bmatrix} C_1 & 0 & 0 & -C_1 & 0 & 0 \\ 0 & 12C_2 & 6LC_2 & 0 & -12C_2 & 6LC_2 \\ 0 & 6LC_2 & 4C_2L^2 & 0 & -6LC_2 & 2C_2L^2 \\ -C_1 & 0 & 0 & C_1 & 0 & 0 \\ 0 & -12C_2 & -6LC_2 & 0 & 12C_2 & -6LC_2 \\ 0 & 6LC_2 & 2C_2L^2 & 0 & -6LC_2 & 4C_2L^2 \end{bmatrix} \quad \begin{aligned} C_1 &= \frac{AE}{L} \\ C_2 &= \frac{EI}{L^3} \end{aligned}$$

$$\hat{f}_{(1)} = \hat{k}\bar{d} = 70 \times 10^3 \begin{bmatrix} 2 & 0 & 0 & -2 & 0 & 0 \\ 0 & 0.067 & 0.10 & 0 & -0.067 & 0.10 \\ 0 & 0.10 & 0.20 & 0 & -0.10 & 0.10 \\ -2 & 0 & 0 & 2 & 0 & 0 \\ 0 & -0.067 & -0.10 & 0 & 0.067 & -0.10 \\ 0 & 0.10 & 0.10 & 0 & -0.10 & 0.20 \end{bmatrix} \begin{Bmatrix} 0.00388 \text{ m} \\ -0.0225 \text{ m} \\ 0.0113 \text{ kN} \cdot \text{m} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

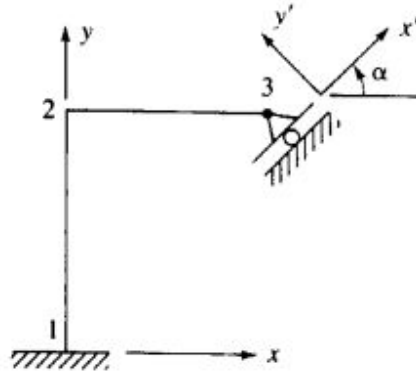
Substituting numerical values into the above equations gives:

$$\begin{Bmatrix} \hat{f}_{1x} \\ \hat{f}_{1y} \\ \hat{m}_1 \\ \hat{f}_{2x} \\ \hat{f}_{2y} \\ \hat{m}_2 \end{Bmatrix} = \begin{Bmatrix} 473 \text{ kN} \\ -26.5 \text{ kN} \\ 0.0 \\ -473 \text{ kN} \\ 26.5 \text{ kN} \\ -78.3 \text{ kN} \cdot \text{m} \end{Bmatrix}$$



## INCLINED OR SKEWED SUPPORTS

If a support is inclined, or skewed, at some angle  $\alpha$  for the global  $x$  axis, as shown below, the boundary conditions on the displacements are not in the global  $x$ - $y$  directions but in the  $x'$ - $y'$  directions.



We must transform the local boundary condition of  $d'_{3y} = 0$  (in local coordinates) into the global  $x$ - $y$  system. Therefore, the relationship between the components of the displacement in the local and the global coordinate systems at node 3 is:

$$\begin{Bmatrix} d'_{3x} \\ d'_{3y} \\ \phi'_3 \end{Bmatrix} = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} d_{3x} \\ d_{3y} \\ \phi_3 \end{Bmatrix}$$

We can rewrite the above expression as:

$$\{d'_3\} = [t_3]\{d_3\} \quad [t_3] = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

We can apply this sort of transformation to the entire displacement vector as:

$$\{d'\} = [T_1]\{d\} \quad \text{or} \quad \{d\} = [T_1]^T\{d'\}$$

where the matrix  $[T_1]$  is:

$$[T_1] = \begin{bmatrix} [I] & [0] & [0] \\ [0] & [I] & [0] \\ [0] & [0] & [t_3] \end{bmatrix}$$

Both the identity matrix  $[I]$  and the matrix  $[t_3]$  are  $3 \times 3$  matrices.

The force vector can be transformed by using the same transformation.

$$\{f'\} = [T_1]\{f\}$$

In global coordinates, the force-displacement equations are:

$$\{f\} = [K]\{d\}$$

Applying the skewed support transformation to both sides of the force-displacement equation gives:

$$[T_i]\{f\} = [T_i][K]\{d\}$$

By using the relationship between the local and the global displacements, the force-displacement equations become:

$$[T_i]\{f\} = [T_i][K][T_i]^T \{d'\} \quad \Rightarrow \quad \{f'\} = [T_i][K][T_i]^T \{d'\}$$

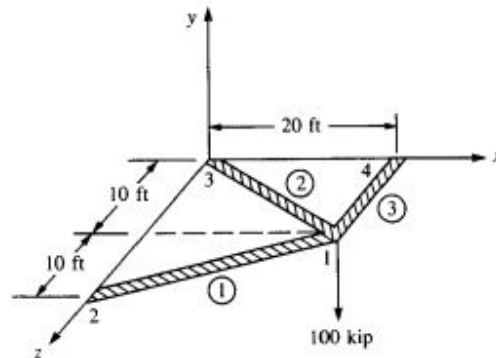
Therefore the global equations become:

$$\begin{Bmatrix} F_x \\ F_y \\ M_1 \\ F_{2x} \\ F_{2y} \\ M_2 \\ F'_{3x} \\ F'_{3y} \\ M_3 \end{Bmatrix} = [T_i][K][T_i]^T \begin{Bmatrix} d_{1x} \\ d_{1y} \\ \phi_1 \\ d_{2x} \\ d_{2y} \\ \phi_2 \\ d'_{3x} \\ d'_{3y} \\ \phi_3 \end{Bmatrix}$$

## FEM GRID EXAMPLES

### EXAMPLE # 1

Consider the frame shown in the figure below.



The frame is fixed at nodes 2, 3, and 4, and is subjected to a load of 100 kips applied at node 1. Assume  $I = 400 \text{ in.}^4$ ,  $J = 110 \text{ in.}^4$ ,  $G = 12 \times 10^3 \text{ ksi}$ , and  $E = 30 \times 10^3 \text{ ksi}$  for all elements.

To facilitate a timely solution, the boundary conditions at nodes 2, 3, and 4 are applied to the local stiffness matrices at the beginning of the solution.

$$\begin{aligned}d_{2y} = \phi_{2x} = \phi_{2z} &= 0 \\d_{3y} = \phi_{3x} = \phi_{3z} &= 0 \\d_{4y} = \phi_{4x} = \phi_{4z} &= 0\end{aligned}$$

#### Beam Element 1:

$$C = \cos \theta = \frac{x_2 - x_1}{L^{(e)}} = \frac{0 - 20}{22.36} = -0.894 \quad S = \sin \theta = \frac{z_2 - z_1}{L^{(e)}} = \frac{20 - 10}{22.36} = 0.447$$

where

$$\frac{12EI}{L^3} = \frac{12(30 \times 10^3)(400)}{(22.36 \times 12)^3} = 7.45 \text{ k/in} \quad \frac{6EI}{L^2} = \frac{6(30 \times 10^3)(400)}{(22.36 \times 12)^2} = 1,000 \text{ k}$$

$$\frac{4EI}{L} = \frac{4(30 \times 10^3)(400)}{(22.36 \times 12)} = 179,000 \text{ k} \cdot \text{in} \quad \frac{GJ}{L} = \frac{(12 \times 10^3)(110)}{(22.36 \times 12)} = 4,920 \text{ k} \cdot \text{in}$$

The global stiffness matrix for element 1, considering only the parts associated with node 1, and the following relationship:

$$k_o = T_o^T \hat{k}_o T_o$$

$$T_o = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & 0.447 \\ 0 & -0.447 & -0.894 \end{bmatrix} \quad T_o^T = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & -0.447 \\ 0 & 0.447 & -0.894 \end{bmatrix}$$

$$\hat{k}^{(e)} = \begin{bmatrix} d_{1y} & \phi_{1x} & \phi_{1z} \\ 7.45 & 0 & 1,000 \\ 0 & 4,920 & 0 \\ 1,000 & 0 & 179,000 \end{bmatrix} \text{ k/in}$$

Therefore, the global stiffness matrix is

$$k^{(1)} = \begin{bmatrix} \phi_{1y} & \phi_{1x} & \phi_{1z} \\ 7.45 & -447 & -894 \\ -447 & 39,700 & 69,600 \\ -894 & 69,600 & 144,000 \end{bmatrix} k/in$$

**Beam Element 2:**

$$C = \cos \theta = \frac{x_3 - x_1}{L^{(2)}} = \frac{0 - 20}{22.36} = -0.894$$

$$S = \sin \theta = \frac{z_3 - z_1}{L^{(2)}} = \frac{0 - 10}{22.36} = -0.447$$

where

$$\frac{12EI}{L^3} = \frac{12(30 \times 10^3)(400)}{(22.36 \times 12)^3} = 7.45 \text{ k/in}$$

$$\frac{6EI}{L^2} = \frac{6(30 \times 10^3)(400)}{(22.36 \times 12)^2} = 1000 \text{ k}$$

$$\frac{4EI}{L} = \frac{4(30 \times 10^3)(400)}{(22.36 \times 12)} = 179,000 \text{ k} \cdot \text{in}$$

$$\frac{GJ}{L} = \frac{(12 \times 10^3)(110)}{(22.36 \times 12)} = 4,920 \text{ k} \cdot \text{in}$$

The global stiffness matrix for element 2, considering only the parts associated with node 1, and the following relationship:

$$k_g = T_g^T \hat{k}_g T_g$$

$$k^{(2)} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & 0.447 \\ 0 & -0.447 & -0.894 \end{bmatrix} \begin{bmatrix} 7.45 & 0 & 1000 \\ 0 & 4,920 & 0 \\ 1,000 & 0 & 179,000 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & -0.447 \\ 0 & 0.447 & -0.894 \end{bmatrix}$$

Therefore, the global stiffness matrix is

$$k^{(2)} = \begin{bmatrix} \phi_{1y} & \phi_{1x} & \phi_{1z} \\ 7.45 & 447 & -894 \\ 447 & 39,700 & -69,600 \\ -894 & -69,600 & 144,000 \end{bmatrix} k/in$$

**Beam Element 3:**

$$C = \cos \theta = \frac{x_4 - x_1}{L^{(3)}} = \frac{20 - 20}{10} = 0$$

$$S = \sin \theta = \frac{z_4 - z_1}{L^{(3)}} = \frac{0 - 10}{10} = -1$$

where

$$\frac{12EI}{L^3} = \frac{12(30 \times 10^3)(400)}{(10 \times 12)^3} = 83.3 \text{ k/in}$$

$$\frac{6EI}{L^2} = \frac{6(30 \times 10^3)(400)}{(10 \times 12)^2} = 5,000 \text{ k}$$

$$\frac{4EI}{L} = \frac{4(30 \times 10^3)(400)}{(10 \times 12)} = 400,000 \text{ k} \cdot \text{in}$$

$$\frac{GJ}{L} = \frac{(12 \times 10^3)(110)}{(10 \times 12)} = 11,000 \text{ k} \cdot \text{in}$$

The global stiffness matrix for element 3, considering only the parts associated with node 1, and the following relationship:

$$k_g = T_g^T \hat{k}_g T_g$$



$$k^{(3)} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix} \begin{bmatrix} 83.3 & 0 & 5,000 \\ 0 & 11,000 & 0 \\ 5,000 & 0 & 400,000 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix}$$

Therefore, the global stiffness matrix is

$$k_{(3)} = \begin{bmatrix} d_{1y} & \phi_{1x} & \phi_{1z} \\ 83.3 & 5,000 & 0 \\ 5,000 & 400,000 & 0 \\ 0 & 0 & 11,000 \end{bmatrix}$$

Superimposing the three elemental stiffness matrices gives:

$$K = \begin{bmatrix} d_{1y} & \phi_{1x} & \phi_{1z} \\ 98.2 & 5,000 & -1,790 \\ 5,000 & 479,000 & 0 \\ -1,790 & 0 & 299,000 \end{bmatrix}$$

The global equations are:

$$\begin{Bmatrix} F_{1y} = -100 \text{ k} \\ M_{1x} = 0 \\ M_{1z} = 0 \end{Bmatrix} = \begin{bmatrix} 98.2 & 5,000 & -1,790 \\ 5,000 & 479,000 & 0 \\ -1,790 & 0 & 299,000 \end{bmatrix} \begin{Bmatrix} d_{1y} \\ \phi_{1x} \\ \phi_{1z} \end{Bmatrix}$$

Solving the above equations gives:

$$\begin{Bmatrix} d_{1y} \\ \phi_{1x} \\ \phi_{1z} \end{Bmatrix} = \begin{Bmatrix} -2.83 \text{ in} \\ 0.0295 \text{ rad} \\ -0.0169 \text{ rad} \end{Bmatrix}$$

**Element 1:** The grid element force-displacement equations can be obtained using  $\hat{f} = \hat{k}_e \bar{T}_e d$ .

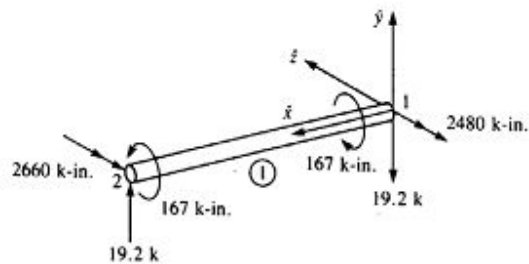
$$\bar{T}_e d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & -0.894 & 0.447 & 0 & 0 & 0 \\ 0 & -0.447 & -0.894 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & -0.894 & 0.447 \\ 0 & 0 & 0 & 0 & -0.447 & -0.894 \end{bmatrix} \begin{bmatrix} -2.83 \text{ in} \\ 0.0295 \text{ rad} \\ -0.0169 \text{ rad} \\ 0 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} -2.83 \text{ in} \\ -0.0339 \text{ rad} \\ 0.00192 \text{ rad} \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(e)} = \hat{k} \bar{T} d = \begin{bmatrix} 7.45 & 0 & 1000 & -7.45 & 0 & 1000 \\ 0 & 4,920 & 0 & 0 & -4,920 & 0 \\ 1000 & 0 & 179,000 & -1000 & 0 & 89,500 \\ -7.45 & 0 & -1,000 & 7.45 & 0 & -1,000 \\ 0 & -4,920 & 0 & 0 & 4,920 & 0 \\ 1000 & 0 & 89,500 & -1,000 & 0 & 179,000 \end{bmatrix} \begin{bmatrix} -2.83 \text{ in} \\ -0.0339 \text{ rad} \\ 0.00192 \text{ rad} \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Solving for the forces and moments gives:

$$\begin{bmatrix} \hat{f}_{1y} \\ \hat{m}_{1x} \\ \hat{m}_{1z} \\ \hat{f}_{2y} \\ \hat{m}_{2x} \\ \hat{m}_{2z} \end{bmatrix} = \begin{bmatrix} -19.2 \text{ k} \\ -167 \text{ k} \cdot \text{in} \\ -2,480 \text{ k} \cdot \text{in} \\ 19.2 \text{ k} \\ 167 \text{ k} \cdot \text{in} \\ -2,260 \text{ k} \cdot \text{in} \end{bmatrix}$$



**Element 2:** The grid element force-displacement equations can be obtained using  $\hat{f} = k_g \bar{T}_g d$ .

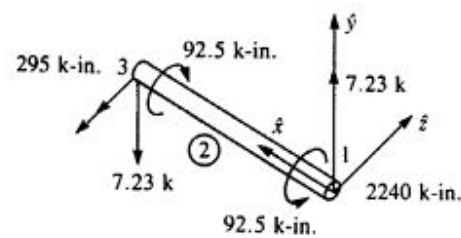
$$\bar{T}_g d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & -0.894 & -0.447 & 0 & 0 & 0 \\ 0 & 0.447 & -0.894 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & -0.894 & -0.447 \\ 0 & 0 & 0 & 0 & 0.447 & -0.894 \end{bmatrix} \begin{Bmatrix} -2.83 \text{ in} \\ 0.0295 \text{ rad} \\ -0.0169 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix} = \begin{Bmatrix} -2.83 \text{ in} \\ -0.0188 \text{ rad} \\ 0.0283 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(e)} = k \bar{T} d = \begin{bmatrix} 7.45 & 0 & 1,000 & -7.45 & 0 & 1,000 \\ 0 & 4,920 & 0 & 0 & -4,920 & 0 \\ 1,000 & 0 & 179,000 & -1,000 & 0 & 89,500 \\ -7.45 & 0 & -1,000 & 7.45 & 0 & -1,000 \\ 0 & -4,920 & 0 & 0 & 4,920 & 0 \\ 1,000 & 0 & 89,500 & -1,000 & 0 & 179,000 \end{bmatrix} \begin{Bmatrix} -2.83 \text{ in} \\ -0.0188 \text{ rad} \\ 0.0283 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Solving for the forces and moments gives:

$$\begin{Bmatrix} \hat{f}_{1y} \\ \hat{m}_{1x} \\ \hat{m}_{1z} \\ \hat{f}_{3y} \\ \hat{m}_{3x} \\ \hat{m}_{3z} \end{Bmatrix} = \begin{Bmatrix} 7.23 \text{ k} \\ -92.5 \text{ k} \cdot \text{in} \\ -2,240 \text{ k} \cdot \text{in} \\ -7.23 \text{ k} \\ 92.5 \text{ k} \cdot \text{in} \\ -295 \text{ k} \cdot \text{in} \end{Bmatrix}$$



**Element 3:** The grid element force-displacement equations can be obtained using  $\hat{f} = k_g \bar{T}_g d$ .

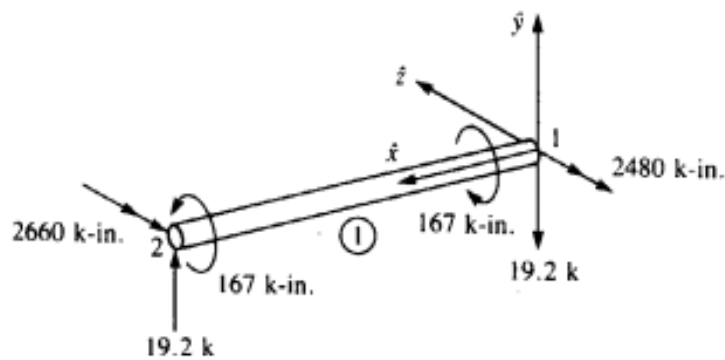
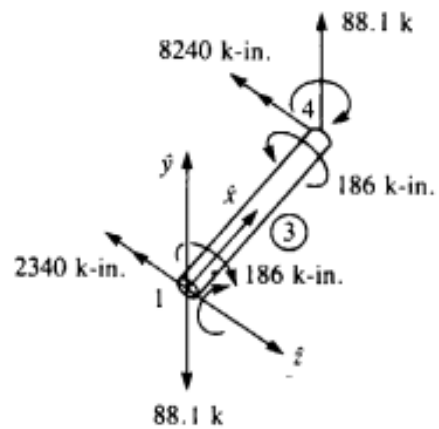
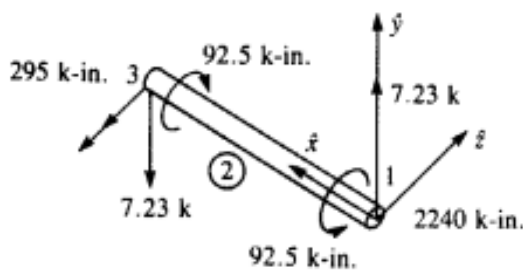
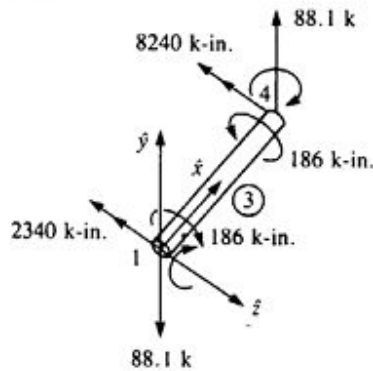
$$\bar{T}_g d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -1 \\ 0 & 0 & 0 & 0 & 1 & 0 \end{bmatrix} \begin{Bmatrix} -2.83 \text{ in} \\ 0.0295 \text{ rad} \\ -0.0169 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix} = \begin{Bmatrix} -2.83 \text{ in} \\ 0.0169 \text{ rad} \\ 0.0295 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(n)} = \hat{k} \bar{T} d = \begin{bmatrix} 83.3 & 0 & 5,000 & -83.3 & 0 & 5,000 \\ 0 & 11,000 & 0 & 0 & -11,000 & 0 \\ 5,000 & 0 & 400,000 & -5,000 & 0 & 200,000 \\ -83.3 & 0 & -5,000 & 83.3 & 0 & -5,000 \\ 0 & -11,000 & 0 & 0 & 11,000 & 0 \\ 5,000 & 0 & 200,000 & -5,000 & 0 & 400,000 \end{bmatrix} \begin{Bmatrix} -2.83 \text{ in} \\ 0.0169 \text{ rad} \\ 0.0295 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Solving for the forces and moments gives:

$$\begin{Bmatrix} \hat{f}_{1y} \\ \hat{m}_{1x} \\ \hat{m}_{1z} \\ \hat{f}_{2y} \\ \hat{m}_{2x} \\ \hat{m}_{2z} \end{Bmatrix} = \begin{Bmatrix} -88.1 \text{ k} \\ 186 \text{ k} \cdot \text{in} \\ -2,340 \text{ k} \cdot \text{in} \\ 88.1 \text{ k} \\ -186 \text{ k} \cdot \text{in} \\ -8,240 \text{ k} \cdot \text{in} \end{Bmatrix}$$



To check the equilibrium of node 1 the local forces and moments for each element need to be transformed to global coordinates. Recall, that:

$$\hat{f} = Tf \Rightarrow f = T^T \hat{f} \quad T^T = T^{-1}$$

Since we are only checking the forces and moments at node 1, we need only the upper-left-hand portion of the transformation matrix  $T_G$ .

Therefore; for **Element 1**:

$$\begin{Bmatrix} f_y \\ m_{1x} \\ m_{1z} \end{Bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & -0.447 \\ 0 & 0.447 & -0.894 \end{bmatrix} \begin{Bmatrix} -19.2 \text{ k} \\ -167 \text{ k} \cdot \text{in} \\ -2,480 \text{ k} \cdot \text{in} \end{Bmatrix} = \begin{Bmatrix} -19.2 \text{ k} \\ 1,260 \text{ k} \cdot \text{in} \\ 2,150 \text{ k} \cdot \text{in} \end{Bmatrix}$$

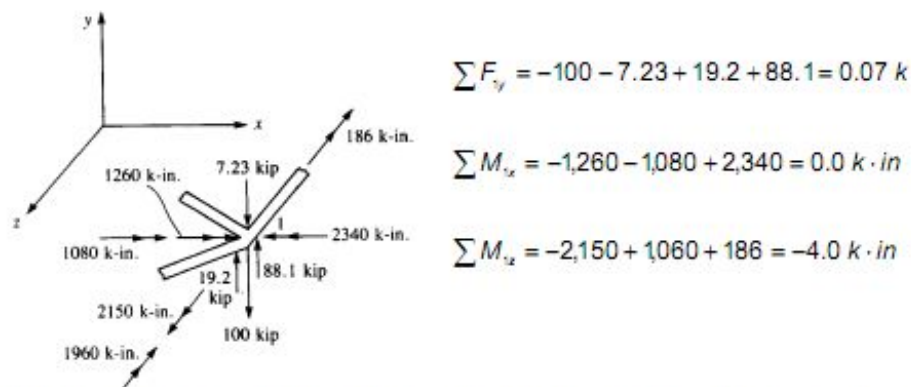
Therefore; for **Element 2**:

$$\begin{Bmatrix} f_y \\ m_{1x} \\ m_{1z} \end{Bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & -0.894 & 0.447 \\ 0 & -0.447 & -0.894 \end{bmatrix} \begin{Bmatrix} -7.23 \text{ k} \\ -92.5 \text{ k} \cdot \text{in} \\ -2,240 \text{ k} \cdot \text{in} \end{Bmatrix} = \begin{Bmatrix} 7.23 \text{ k} \\ 1,080 \text{ k} \cdot \text{in} \\ -1,960 \text{ k} \cdot \text{in} \end{Bmatrix}$$

Therefore; for **Element 3**:

$$\begin{Bmatrix} f_y \\ m_{1x} \\ m_{1z} \end{Bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix} \begin{Bmatrix} -88.1 \text{ k} \\ -2,340 \text{ k} \cdot \text{in} \\ -186 \text{ k} \cdot \text{in} \end{Bmatrix} = \begin{Bmatrix} -88.1 \text{ k} \\ -2,340 \text{ k} \cdot \text{in} \\ -186 \text{ k} \cdot \text{in} \end{Bmatrix}$$

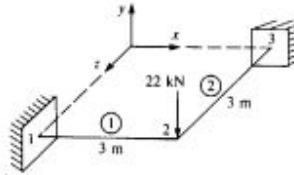
The forces and moments that are applied to node 1 by each element are equal in magnitude and opposite direction. Therefore the sum of the forces and moments acting on node 1 are:



The forces and moments accurately satisfy equilibrium considering the amount of truncation error inherent in results of the calculations presented in this example.

## EXAMPLE # 2

Consider the frame shown in the figure below.



The frame is fixed at nodes 1 and 3, and is subjected to a load of 22 kN applied at node 2. Assume  $I = 16.6 \times 10^{-5} \text{ m}^4$ ,  $J = 4.6 \times 10^{-5} \text{ m}^4$ ,  $G = 84 \text{ GPa}$ , and  $E = 210 \text{ GPa}$  for all elements.

To facilitate a timely solution, the boundary conditions at nodes 1 and 3 are applied to the local stiffness matrices at the beginning of the solution.

$$d_{1y} = \phi_{1x} = \phi_{1z} = 0$$

$$d_{3y} = \phi_{3x} = \phi_{3z} = 0$$

**Beam Element 1:** the local  $x$  axis coincides with the global  $x$  axis

$$C = \cos \theta = \frac{x_2 - x_1}{L^{(1)}} = \frac{3}{3} = 1 \quad S = \sin \theta = \frac{z_2 - z_1}{L^{(1)}} = \frac{0}{3} = 0$$

where

$$\frac{12EI}{L^3} = \frac{12(210 \times 10^9)(16.6 \times 10^{-5})}{(3)^3} = 1.55 \times 10^4 \text{ kN/m}$$

$$\frac{6EI}{L^2} = \frac{6(210 \times 10^9)(16.6 \times 10^{-5})}{(3)^2} = 2.32 \times 10^4 \text{ kN}$$

$$\frac{4EI}{L} = \frac{4(210 \times 10^9)(16.6 \times 10^{-5})}{3} = 4.65 \times 10^4 \text{ kN-m}$$

$$\frac{GJ}{L} = \frac{(84 \times 10^9)(4.6 \times 10^{-5})}{3} = 0.128 \times 10^4 \text{ kN-m}$$

The global stiffness matrix for element 1, considering only the parts associated with node 2, may be obtained from the following relationship:

$$k_g = T_g^T \hat{k}_g T_g$$

$$k^{(1)} = 10^4 \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1.55 & 0 & -2.32 \\ 0 & 0.128 & 0 \\ -2.32 & 0 & 4.65 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \text{ kN/m}$$

Therefore, the global stiffness matrix is

$$k_{(1)} = 10^4 \begin{bmatrix} d_{2y} & \phi_{2x} & \phi_{2z} \\ 1.55 & 0 & -2.32 \\ 0 & 0.128 & 0 \\ -2.32 & 0 & 4.65 \end{bmatrix} \text{ kN/m}$$

**Beam Element 2:** the local  $\hat{x}$  axis is located from node 2 to node 3

$$C = \cos\theta = \frac{x_3 - x_2}{L^{(2)}} = \frac{0}{3} = 0 \quad S = \sin\theta = \frac{z_3 - z_2}{L^{(1)}} = \frac{-3}{3} = -1$$

The global stiffness matrix for element 2, considering only the parts associated with node 2, may be obtained using:

$$k_s = T_s^T \hat{k}_s T_s$$

$$k^{(2)} = 10^4 \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix} \begin{bmatrix} 1.55 & 0 & -2.32 \\ 0 & 0.128 & 0 \\ -2.32 & 0 & 4.65 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & -1 \\ 0 & 1 & 0 \end{bmatrix} \text{ kN/m}$$

Therefore, the global stiffness matrix is

$$k_{(2)} = 10^4 \begin{bmatrix} d_{2y} & \phi_{2x} & \phi_{2z} \\ 1.55 & 2.32 & 0 \\ 2.32 & 4.65 & 0 \\ 0 & 0 & 0.128 \end{bmatrix} \text{ kN/m}$$

Superimposing the two elemental stiffness matrices gives:

$$K = 10^4 \begin{bmatrix} d_{2y} & \phi_{2x} & \phi_{2z} \\ 3.10 & 2.32 & -2.32 \\ 2.32 & 4.78 & 0 \\ -2.32 & 0 & 4.78 \end{bmatrix} \text{ kN/m}$$

The global equations are:

$$\begin{Bmatrix} F_{2y} = -22 \text{ kN} \\ M_{2x} = 0 \\ M_{2z} = 0 \end{Bmatrix} = 10^4 \begin{bmatrix} 3.10 & 2.32 & -2.32 \\ 2.32 & 4.78 & 0 \\ -2.32 & 0 & 4.78 \end{bmatrix} \begin{Bmatrix} d_{2y} \\ \phi_{2x} \\ \phi_{2z} \end{Bmatrix}$$

Solving the above equations gives:

$$\begin{Bmatrix} d_{2y} \\ \phi_{2x} \\ \phi_{2z} \end{Bmatrix} = \begin{Bmatrix} -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ -0.00126 \text{ rad} \end{Bmatrix}$$

**Element 1:** The grid element force-displacement equations can be obtained using  $\hat{f} = \hat{k}_g \bar{T}_g d$ .

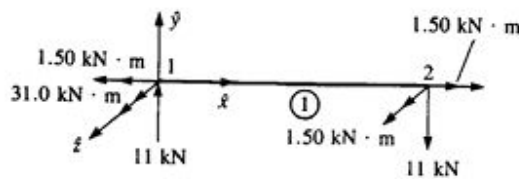
$$\bar{T}_g d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ 0 \\ -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ -0.00126 \text{ rad} \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ -0.00126 \text{ rad} \end{Bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(1)} = \hat{k}\bar{T}d = 10^4 \begin{bmatrix} 1.55 & 0 & 2.32 & -1.55 & 0 & 2.32 \\ 0 & 0.128 & 0 & 0 & -0.128 & 0 \\ 2.32 & 0 & 4.65 & -2.32 & 0 & 2.33 \\ -1.55 & 0 & -2.32 & 1.55 & 0 & -2.32 \\ 0 & -0.128 & 0 & 0 & 0.128 & 0 \\ 2.32 & 0 & 2.33 & -2.32 & 0 & 4.65 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ 0 \\ -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ -0.00126 \text{ rad} \end{Bmatrix}$$

Solving for the forces and moments gives:

$$\begin{Bmatrix} \hat{f}_{1y} \\ \hat{m}_{1,x} \\ \hat{m}_{1,z} \\ \hat{f}_{2y} \\ \hat{m}_{2,x} \\ \hat{m}_{2,z} \end{Bmatrix} = \begin{Bmatrix} 11.0 \text{ kN} \\ -1.50 \text{ kN} \cdot \text{m} \\ 31.0 \text{ kN} \cdot \text{m} \\ -11.0 \text{ kN} \\ 1.50 \text{ kN} \cdot \text{m} \\ 1.50 \text{ kN} \cdot \text{m} \end{Bmatrix}$$



**Element 2:** The grid element force-displacement equations can be obtained using  $\hat{f} = \hat{k}_e \bar{T}_e d$ .

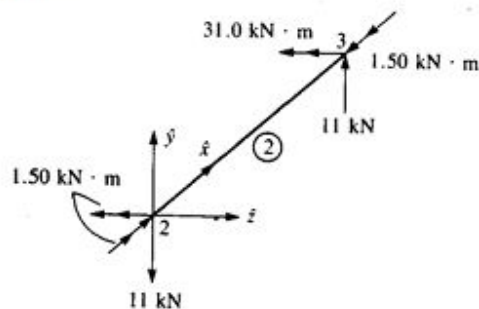
$$\bar{T}_e d = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & -1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -1 \\ 0 & 0 & 0 & 0 & 1 & 0 \end{bmatrix} \begin{Bmatrix} -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ -0.00126 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix} = \begin{Bmatrix} -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ 0.00126 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Therefore, the local force-displacement equations are:

$$\hat{f}_{(2)} = \hat{k}\bar{T}d = 10^4 \begin{bmatrix} 1.55 & 0 & 2.32 & -1.55 & 0 & 2.32 \\ 0 & 0.128 & 0 & 0 & -0.128 & 0 \\ 2.32 & 0 & 4.65 & -2.32 & 0 & 2.33 \\ -1.55 & 0 & -2.32 & 1.55 & 0 & -2.32 \\ 0 & -0.128 & 0 & 0 & 0.128 & 0 \\ 2.32 & 0 & 2.33 & -2.32 & 0 & 4.65 \end{bmatrix} \begin{Bmatrix} -0.00259 \text{ m} \\ 0.00126 \text{ rad} \\ 0.00126 \text{ rad} \\ 0 \\ 0 \\ 0 \end{Bmatrix}$$

Solving for the forces and moments gives:

$$\begin{Bmatrix} \hat{f}_{2y} \\ \hat{m}_{2,x} \\ \hat{m}_{2,z} \\ \hat{f}_{3y} \\ \hat{m}_{3,x} \\ \hat{m}_{3,z} \end{Bmatrix} = \begin{Bmatrix} -11.0 \text{ kN} \\ 1.50 \text{ kN} \cdot \text{m} \\ -1.50 \text{ kN} \cdot \text{m} \\ 11.0 \text{ kN} \\ -1.50 \text{ kN} \cdot \text{m} \\ -31.0 \text{ kN} \cdot \text{m} \end{Bmatrix}$$





The resulting free-body diagrams:

