Multi degree of freedom (MDOF) systems are usually analyzed using Modal Analysis. A typical MDOF system with 'n' degree of freedom is shown in Figure 1 . This system when subjected to ground motion undergoes deformations in number of possible ways. These deformed shapes are known as modes of vibration or mode shapes. Each shape is vibrating with a particular natural frequency. Total unique modes for each MDOF system are equal to the possible degree of freedom of system. The equations of motion for MDOF system is given by

 $[m]\{\ddot{x}(t)\} + [c]\{\dot{x}(t)\} + [k]\{\dot{x}(t)\} = - [m]\{r\} \ddot{x}_{g}(t)$

MDOF system with 'n' degrees-of-freedom.

 $[m]\{x(t)\} + [c]\{x(t)\} + [k]\{x(t)\} = - [m]\{r\}\ \ddot{x}_{g}(t)$

where, $[m]$ = Mass matrix $(n \times n)$; $[k]$ = Stiffness matix $(n \times n)$; $[c]$ = Damping matrix $(n \times n)$ n); $\{r\}$ = Influence coefficient vector $(n\times 1)$; $\{x(t)\}$ = relative displacement vector; $\{\dot{x}(t)\}$ = relative velocity vector, $\{\ddot{x}(t)\}$ = relative acceleration vector, and $\ddot{x}_g(t)$ = earthquake ground acceleration.

The undamped eigen values and eigen vectors of the MDOF system are found form the characteristic equation $\{ [k] - \omega_i^2[m] \} \phi_i = 0$ $i = 1, 2, 3, ..., n$

$$
\det \left| \left\{ \left[k \right] - \omega_i^2 \left[m \right] \right\} \right| = 0
$$

where,

 ω_i^2 = eigen values of the ith mode ϕ_i = eigen vector or mode shape of the ith mode ω_i = natural frequency in the ith mode.

Let the displacement response of the MDOF system is expressed as

 $\{x(t)\} = [\phi] \{y(t)\}$

where $\{y(t)\}\$ represents the modal displacement vector, and $\left[\phi\right]$ is the mode shape matrix given by

1

 $[\phi] = [\phi_1, \phi_2, \dots, \phi_n]$

Substituting $\{x\} = [\phi]\{y\}$ in equation (4.18) and pre-multiply by $[\phi]$ ^T

 $\left[\phi\right]^T \left[m\right] \left[\phi\right] \left\{\ddot{y}(t)\right\} + \left[\phi\right]^T \left[c\right] \left[\phi\right] \left\{\dot{y}(t)\right\} + \left[\phi\right]^T \left[k\right] \left[\phi\right] \left\{y(t)\right\} = -\left[\phi\right]^T \left[m\right] \left\{r\right\} \ddot{x}_e(t)$

The above equation reduces to

 $[M_m]\{\ddot{y}(t)\} + [C_d]\{\dot{y}(t)\} + [K_d]\{\dot{y}(t)\} = -[\phi]^T[m]\{r\} \ddot{x}_e(t)$

where.

 $\left[\phi\right]^T[m][\phi]=[M_m]=$ generalized mass matrix $\left[\phi\right]^{T}\left[c\right]\left[\phi\right]=\left[C_{d}\right]$ = generalized damping matrix $\left[\phi\right]^T [k] [\phi] = [K_d]$ = generalized stiffness matrix

By virtue of the properties of the $\lceil \phi \rceil$, the matrices $\lceil M_m \rceil$ and $\lceil K_d \rceil$ are diagonal matrices. However, for the classically damped system (i.e. if the $[C_d]$ is also a diagonal matrix), the equation (1) reduces to the following equation

$$
\ddot{y}_i(t) + 2\xi_i \omega_i \dot{y}_i(t) + \omega_i^2 y_i(t) = -\Gamma_i \ddot{x}_g(t) \qquad (i = 1, 2, 3, ..., n)
$$

where,

$$
y_i(t) = \text{modal displacement response in the i}^{\text{th}} \text{mode},
$$

$$
\xi_i = \text{modal damping ration in the i}^{\text{th}} \text{mode, and}
$$

$$
\Gamma_i = \text{modal participation factor for i}^{\text{th}} \text{mode expressed by}
$$

$$
\Gamma_i = \frac{\{\phi_i\}^T[m]\{r\}}{\{\phi_i\}^T[m]\{\phi_i\}}
$$

The maximum displacement response of the structure in the ith mode is

$$
x_{i,\max} = \phi_i \Gamma_i S_d \quad (\xi_i, \omega_i) \tag{1} \tag{1} \tag{1} \tag{1}
$$

The maximum acceleration response of the structure in the ith mode is

$$
\left\{\ddot{x}_a\right\}_{i,\text{max}} = \left\{\phi_i\right\} \Gamma_i S_{pa}(\xi_i, \omega_i) \tag{i = 1, 2, \dots, n}
$$

The required response quantity of interest, r_i i.e. (displacement, shear force, bending moment etc.) of the structure can be obtained in each mode of vibration using the maximum response obtained in equations. However, the final maximum response, r_{max} shall be obtained by combining the response in each mode of vibration using the modal combinations rules.

Modal Combination Rules

The commonly used methods for obtaining the peak response quantity of interest for a MDOF system are as follows:

- Absolute Sum (ABSSUM) Method,
- Square root of sum of squares (SRSS) method, and
- Complete quadratic combination (CQC) method

In ABSSUM method, the peak responses of all the modes are added algebraically, assuming that all modal peaks occur at same time. The maximum response is given by

$$
r_{\max} = \sum_{i=1}^{n} |r_i|
$$

In the SRSS method, the maximum response is obtained by square root of sum of square of response in each mode of vibration and is expressed by

$$
r_{\max} = \sqrt{\sum_{i=1}^{n} r_i^2}
$$

Response spectra of El-Centro, 1940 earthquake ground motion.

A three-story building is modeled as 3-DOF system and rigid floors as shown in Figure Determine the top floor maximum displacement and base shear due to El-Centro, 1940 earthquake ground motion using the response spectrum method. Take the inter-story lateral stiffness of floors i.e. $k_1 = k_2 = k_3 = 16357.5 \times 10^3$ N/m and the floor mass $m_1 = m_2 = 10000$ kg and $m_3 = 5000$ kg.

$$
\begin{bmatrix} m \\ m \end{bmatrix} = \begin{bmatrix} m_1 & 0 & 0 \\ 0 & m_2 & 0 \\ 0 & 0 & m_3 \end{bmatrix} = \begin{bmatrix} 10000 & 0 & 0 \\ 0 & 10000 & 0 \\ 0 & 0 & 5000 \end{bmatrix}
$$

and the stiffness matrix,

$$
\begin{bmatrix} k \end{bmatrix} = \begin{bmatrix} k_1 + k_2 & -k_2 & 0 \\ -k_2 & k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 \end{bmatrix} = \begin{bmatrix} 32715 & -16357.5 & 0 \\ -16357.5 & 32715 & -16357.5 \\ 0 & -16357.5 & 16357.5 \end{bmatrix}
$$

Finding eigen values and eigen vectors using the equation

 $\left\{ \left[k\right] -\varpi ^{2}\left[m\right] \right\} \left\{ \varphi \right\} =0$

det $\left[\left[k\right]-\omega^2\left[m\right]\right]=0$

Example 2

$$
\det \begin{bmatrix} 2 & -1 & 0 \\ 16357.5 \times 10^3 \begin{bmatrix} 2 & -1 & 0 \\ -2 & 2 - 2\lambda & -1 \\ 0 & -1 & 1 - \lambda \end{bmatrix} - \omega^2 \times 5000 \begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} = 0
$$

\n
$$
\lambda = \frac{\omega^2 \times 5000}{16357.5 \times 10^3}
$$

\n
$$
\det \begin{vmatrix} 2 - 2\lambda & -1 & 0 \\ -1 & 2 - 2\lambda & -1 \\ 0 & -1 & 1 - \lambda \end{vmatrix} = 0
$$

\n
$$
(2 - 2\lambda) \Big[(2 - 2\lambda)(1 - \lambda) - 1 \Big] + [-1 + \lambda] = 0
$$

\n
$$
(2 - 2\lambda) \Big[2 - 2\lambda - 2\lambda + 2\lambda^2 - 1 \Big] - 1 + \lambda = 0
$$

\n
$$
2 - 2\lambda \Big[2\lambda^2 - 4\lambda + 1 \Big] + \lambda - 1 = 0
$$

\n
$$
4\lambda^2 - 8\lambda + 2 - 4\lambda^3 + 2\lambda^2 - 2\lambda + \lambda - 1 = 0
$$

\n
$$
-4\lambda^3 + 12\lambda^2 - 9\lambda + 1 = 0
$$

\n
$$
\lambda_1 = 0.134, \quad \lambda_2 = 1, \quad \lambda_3 = 1.866
$$

Implying that

$$
\omega_1 = 20.937 \text{ rad/sec}
$$
 $\omega_2 = 57.2 \text{ rad/sec}$ $\omega_3 = 78.13 \text{ rad/sec}$

On substituting ω^2 in the characteristic equation,

For Mode 1

$$
\begin{bmatrix} 2-2 \times 0.134 & -1 & 0 \\ -1 & 2-2 \times 0.134 & -1 \\ 0 & -1 & 1-0.134 \end{bmatrix} \begin{bmatrix} \phi_{11} \\ \phi_{21} \\ \phi_{31} \end{bmatrix} = 0
$$

Assuming $\Phi_{11} = 1$,

 $2 - 2 \times 0.134 = \phi_{21}$

$$
\phi_{21} = 1.732 \text{ and } \phi_{31} = 2.0
$$

$$
\{\phi_1\} = \begin{cases} 1\\ 1.732\\ 2.0 \end{cases}
$$

For Mode 2

$$
\begin{bmatrix} 2-2 & -1 & 0 \\ -1 & 2-2 & -1 \\ 0 & -1 & 1-1 \end{bmatrix} \begin{bmatrix} \phi_{12} \\ \phi_{22} \\ \phi_{32} \end{bmatrix} = 0
$$

$$
\{\varphi_2\} = \begin{Bmatrix} 1 \\ 0 \\ 1 \end{Bmatrix}
$$

For Mode 3

$$
\begin{bmatrix} 2-2 \times 1.866 & -1 & 0 \\ -1 & 2-2 \times 1.866 & -1 \\ 0 & -1 & 1-1.866 \end{bmatrix} \begin{bmatrix} \phi_{13} \\ \phi_{23} \\ \phi_{33} \end{bmatrix} = 0
$$

$$
\left\{\varphi_3\right\} = \begin{cases} 1 \\ -1.733 \\ 2.0 \end{cases}
$$

$$
\Gamma_{1} = \frac{\{\phi_{1}\}^{T} \{m\} \{r\}}{\{\phi_{1}\}^{T} \{m\} \{\phi\}} = \frac{\begin{bmatrix} 1 & 1.733 & 2 \end{bmatrix} \times 5000 \begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}}{\begin{bmatrix} 1 & 1.733 & 2 \end{bmatrix} \times 5000 \begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 1.733 \\ 2 \end{bmatrix}}
$$

 $\Gamma_1 = 0.622$

Similarly,

$$
\Gamma_2 = \frac{\{\phi_2\}^T \{m\} \{r\}}{\{\phi_2\}^T \{m\} \{\phi_2\}} = 0.333
$$

$$
\Gamma_3 = \frac{\left\{\phi_2\right\}^T \left\{m\right\} \left\{r\right\}}{\left\{\phi_3\right\}^T \left\{m\right\} \left\{\phi\right\}} = 0.045
$$

1st Mode Response

$$
T_1 = \frac{2\pi}{\omega_1} = \frac{2\pi}{20.937} = 0.30 \text{ sec}
$$

 $\xi_1 = 0.02$

From the response spectra

 $S_{d1} = 0.01902m$

Base shear

Top floor displacement = $\Gamma_1 \times \phi_{31} \times S_{d1} = 2 \times 0.622 \times 0.01902$

 $= 0.0236m$ = $k \times \phi_{11} \times \Gamma_1 \times S_{d1} = 16.357 \times 10^6 \times 1 \times 0.622 \times 0.01902$ $= 193 kN$

2^{nd} Mode Response

$$
T_2 = \frac{2\pi}{57.2} = 0.11 \text{ sec}
$$

 $\xi_2 = 0.02$

from response spectra

 $S_{d2} = 0.00231m$

Top floor displacement = $\phi_{32} \times \Gamma_2 \times S_{d2} = -1 \times 0.333 \times 0.00231$

$$
=-7.69\times10^{-4}m
$$

Base shear

 $k = k \times \phi_{12} \times \Gamma_2 \times S_{d2} = 16357 \times 10^3 \times 1 \times 0.333 \times 0.00231 = 12.58kN$

3^{rd} Mode Response

$$
T_3 = \frac{2\pi}{78.13} = 0.08 \sec
$$

 $\xi_3 = 0.02$

from response spectra

 $S_{d3} = 9.77 \times 10^{-4} m$ Top floor displacement = $\phi_{33} \times \Gamma_3 \times S_{d3}$ $= 2 \times 0.045 \times 9.77 \times 10^{-4} m$ $= 8.793 \times 10^{-5} m$ $= k \times \phi_{13} \times \Gamma_3 \times S_{d3}$ Base shear $= 16357.5 \times 10^{3} \times 1 \times 0.045 \times 9.77 \times 10^{-4}$ $= 0.719 kN$

Peak responses using the SRSS modal combination rule are given below

ASSIGNMENT

A two-story building in Lahore is modeled as 2-DOF system and rigid floors on stiff soil as shown in the Figure. Determine the top floor maximum displacement and base shear using UBC-97 response spectra. Take the inter-story stiffness, *k =200× 10³ N/m and the floor mass, m = 2500 kg.*

UBC Design Response Spectra

FIGURE 16-3-DESIGN RESPONSE SPECTRA

TABLE 16-I-SEISMIC ZONE FACTOR Z

NOTE: The zone shall be determined from the seismic zone map in Figure 16-2.

TABLE 16-J-SOIL PROFILE TYPES

¹Soil Profile Type S_E also includes any soil profile with more than 10 feet (3048 mm) of soft clay defined as a soil with a plasticity index, $PI > 20$, $w_{mc} \ge 40$ percent and $s_u < 500$ psf (24 kPa). The Plasticity Ind

TABLE 16-Q-SEISMIC COEFFICIENT Ca

¹Site-specific geotechnical investigation and dynamic site response analysis shall be performed to determine seismic coefficients for Soil Profile Type S_F

TABLE 16-R-SEISMIC COEFFICIENT Cv

¹Site-specific geotechnical investigation and dynamic site response analysis shall be performed to determine seismic coefficients for Soil Profile Type SF.